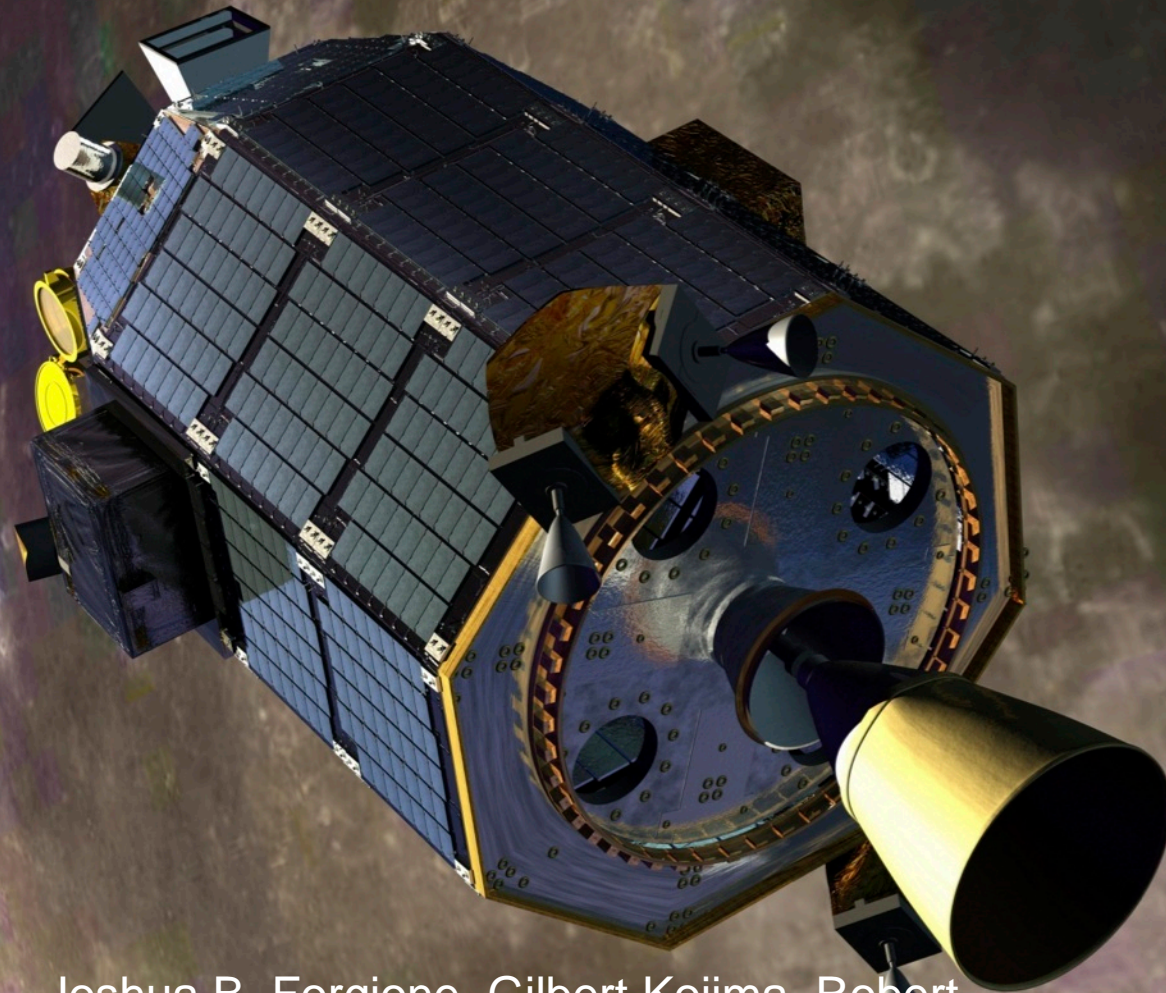


Low-Cost, Risk-Reduction Testing of Class D Spacecraft Photovoltaic Systems

The LADEE Approach



Joshua B. Forgione, Gilbert Kojima, Robert
Hanel, Mark Mallinson, Joseph Camisa



Contents

- LADEE** Background & Power System
- Large Area Pulsed Solar Simulation (LAPSS) & Risk
- The LADEE Solution
- Conclusions

The Bottom Line

A low-cost, short-lifetime, Class D mission eliminated LAPSS testing due to resource constraints. The team instead developed a simple, low-cost, COTS-based method to verified power generation requirements and reduce technical risk.

****LADEE – Lunar Atmospheric and Dust Environment Explorer**

Background – LADEE Mission

Objective

- Measure Lunar Dust
- Examine the Lunar atmosphere

Key parameters

- Launched Sept 6, 2013
- Science Data Acquisition: 100 days
- 1 Month Mission Extension (Impact ~4/21/14)

Spacecraft

- Type: Small Orbiter - Category II, Enhanced Class D
- Provider: ARC/GSFC

Instruments

- Science Instruments: NMS, UVS, and LDEX
- Technology Payload: Lunar Laser Communications Demo

Launch Vehicle: Minotaur V

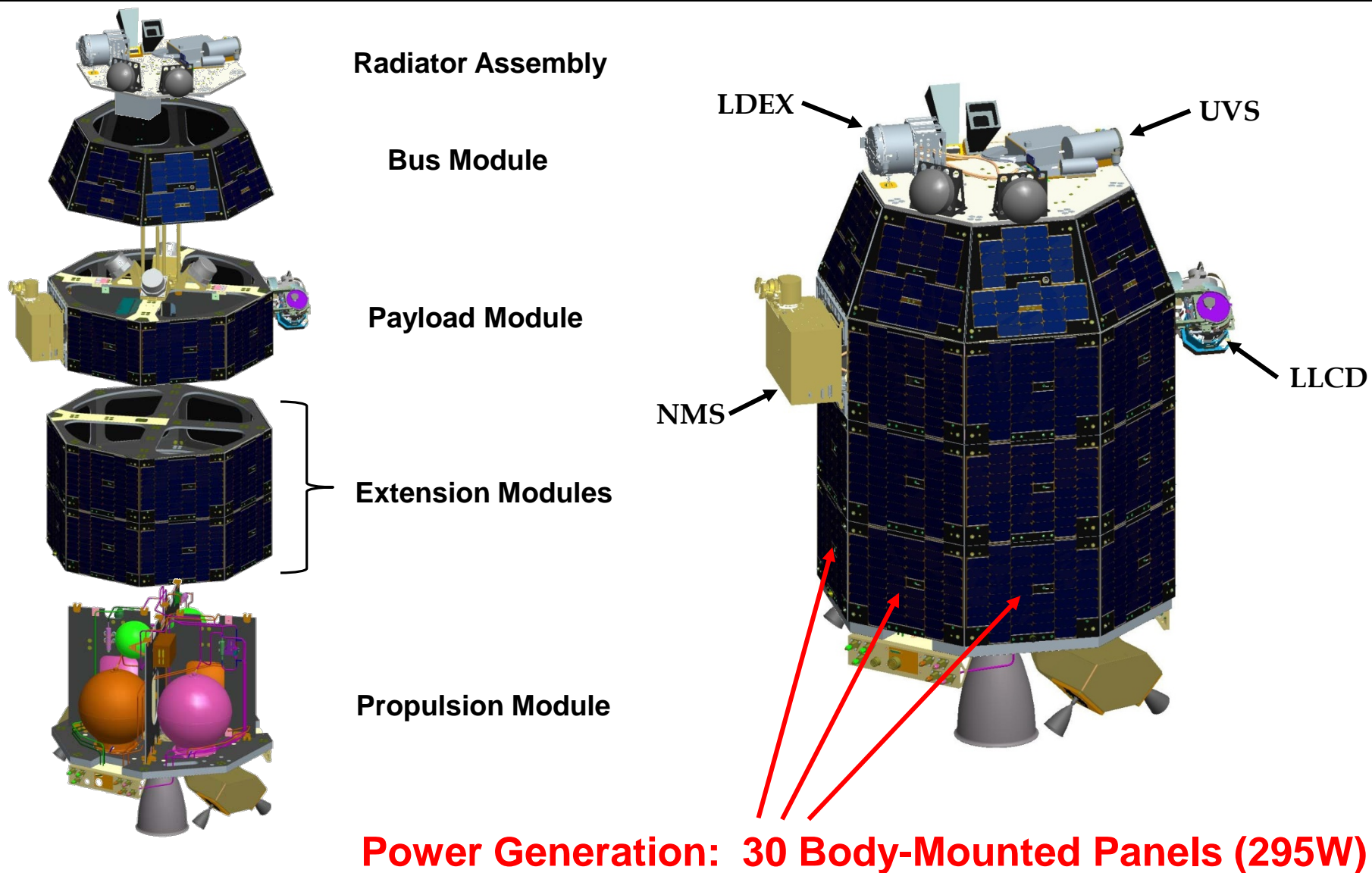
Launch Site: Wallops Flight Facility (WFF)

Class D: Low complexity, high-risk, short lifetime
(NPR 8704.5)



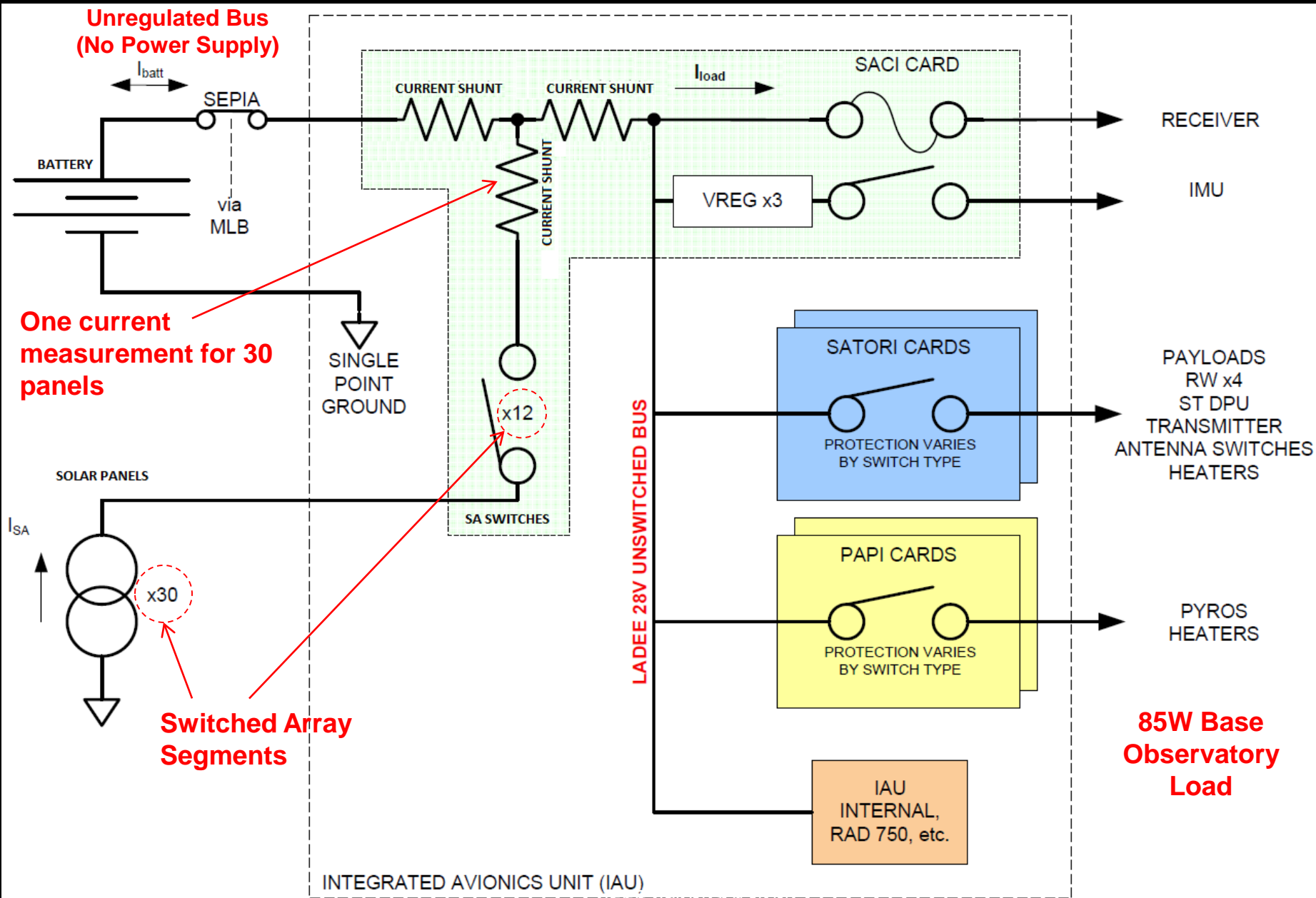
Integrated LADEE @ WFF

LADEE Modular Spacecraft Bus





LADEE Electrical Power System (EPS)





Large Area Pulsed Solar Simulation

LAPSS is the 'Gold Standard' for aerospace array testing

Does it make sense for LADEE?

PRO

Very accurate (<2%)
End-to-End Calibration
Safe – pulsed light does not heat array
Can detect minor defects
Facilities exist within NASA

CON

ARC has no LAPSS – either build one or test off-site
Expensive (>\$100k)
Lengthy (test or build)
Complex
Heavy – 900kg (~1 ton)

Want to know more about LAPSS?

- Mueller, R.L. *The Large Area Pulsed Solar Simulator*. NASA Technical Report #NASA-CR-194507, Aug 15. 1993
- <http://www.spectrolab.com/simulators.htm>

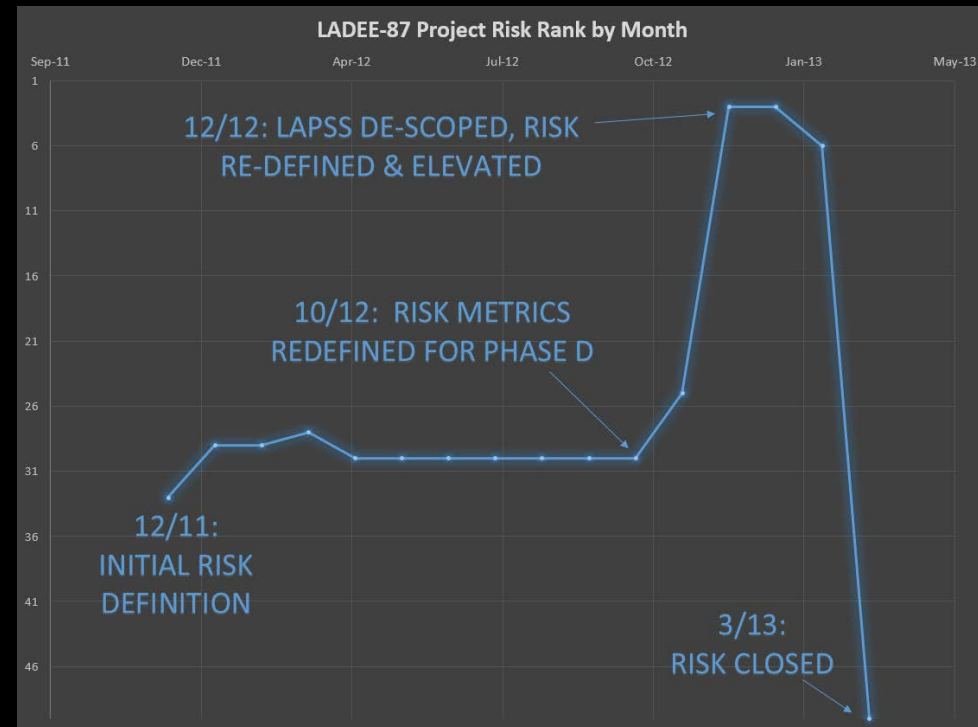


LAPSS and Risk

- Enter Phase D, LADEE faced substantial schedule challenges and redefined its risk metrics.
- The original risk (LADEE-87), related to scheduling of the LAPSS test
- In Phase D, LAPSS testing was cancelled – the risk morphed & elevated

Given that: LAPSS testing was only performed for each individual solar panel by the vendor.

There is a possibility that: Without Observatory level LAPSS system performance testing for the interconnected solar panels, which are combined through the Primary cable harness panel interconnections, the requirements verification of the 295 Watts solar panel output power will not be verified by testing.



The LADEE Risk Management process is not novel, and typical of NASA programs

Class D resource constraints forced the project to take a risk & innovate

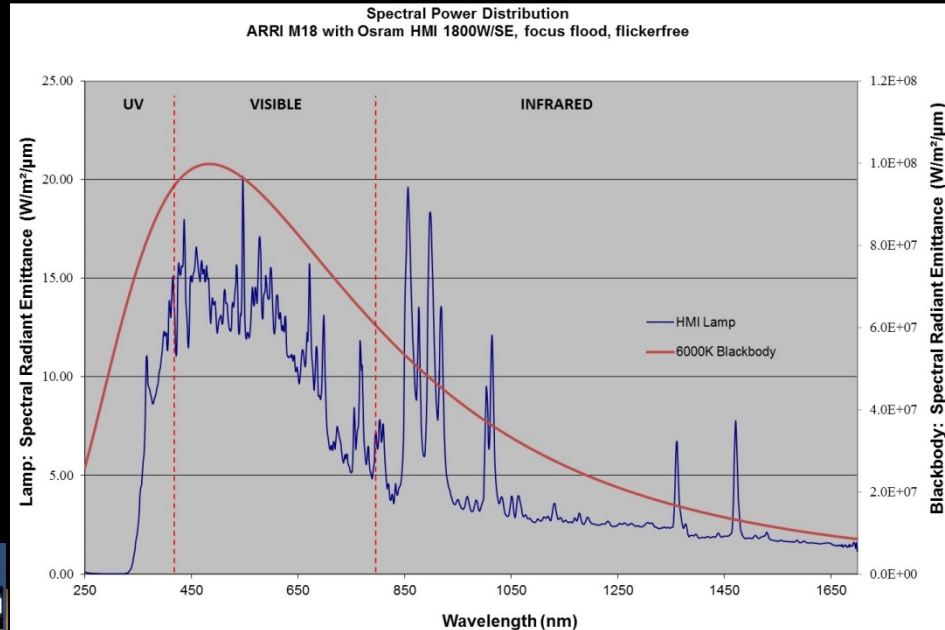
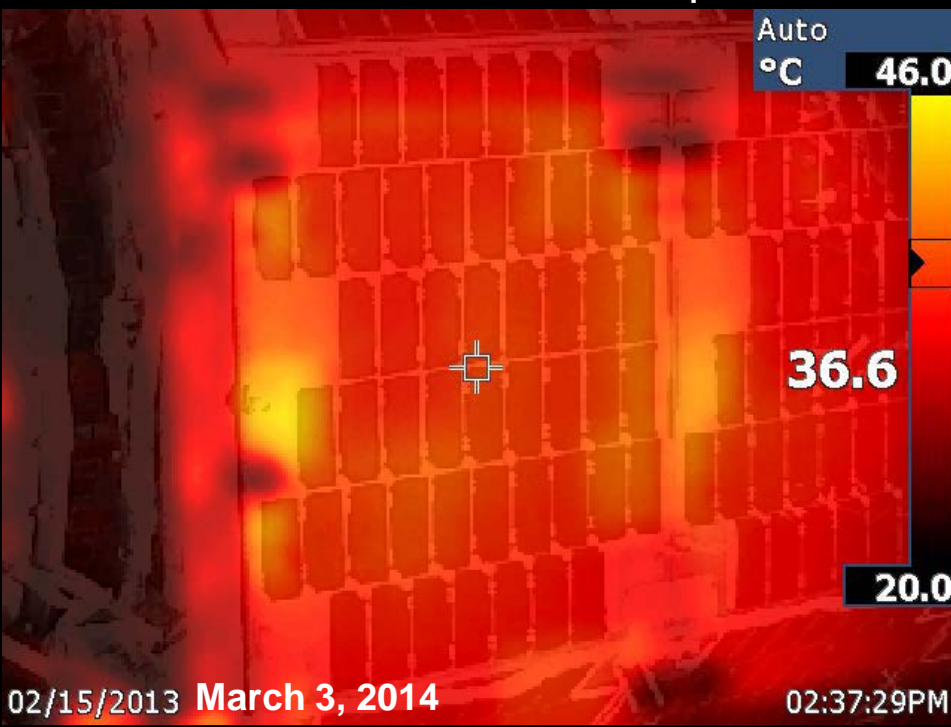
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Developing the Solution

Key Requirements:

- Generate rated power (295W req.)
- Hardware safety (temperature)
- Personnel safety (UV)
- Quick test (<1 day)
- Low Cost

Research & Experimentation yielded a COTS solution that exceeds requirements





Observatory Test

295W req. verified!

Thermal imagers
monitor temperature

ANSI z87.1 glasses
protect from UV

Test repeated 3
times before launch

Easy to perform
inside 10k clean tent

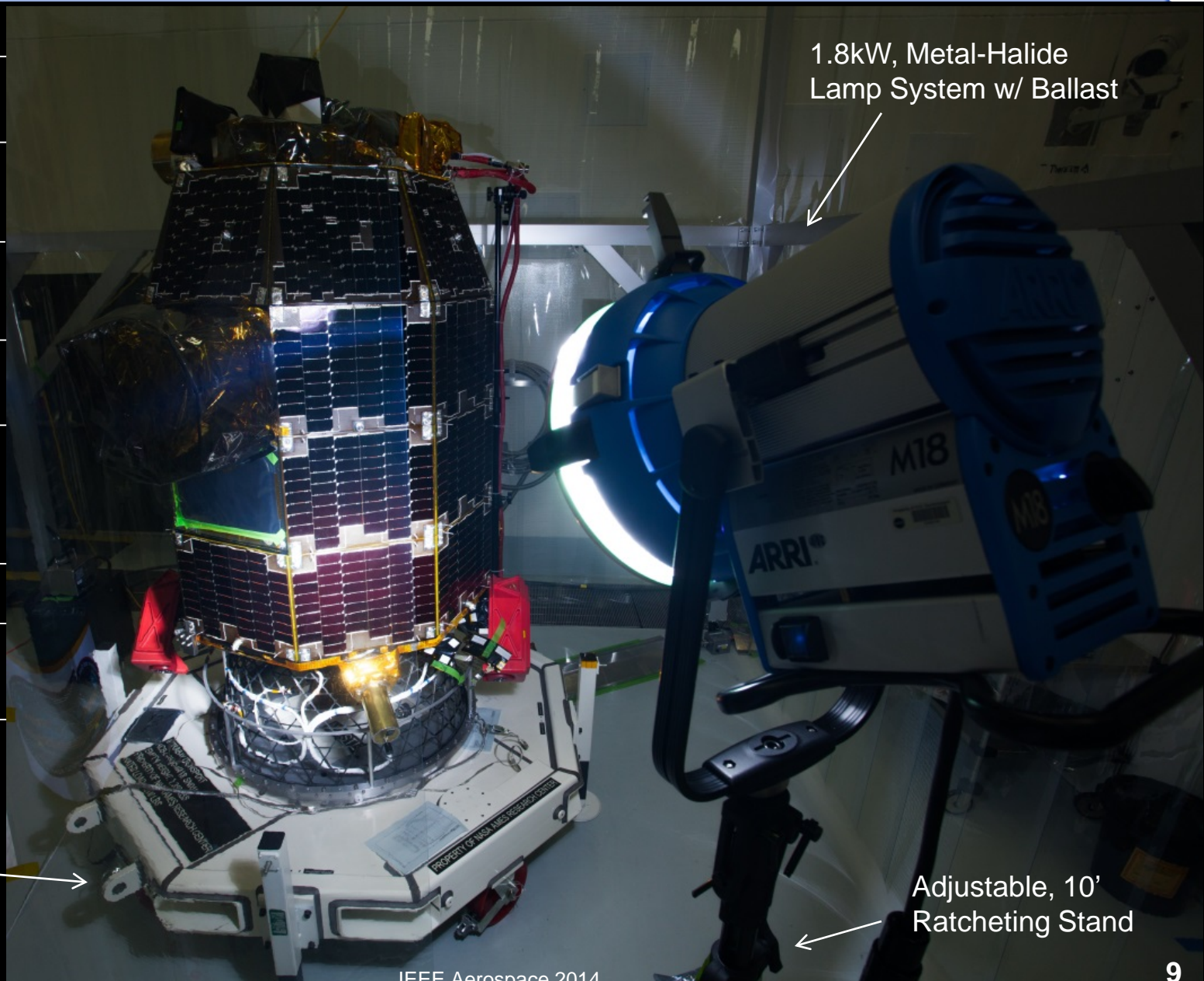
Lamp System Cost:
\$10k Buy
\$750/wk Rent

Test Time: 4 Hours

System weight:
~45kg (~100lb)

‘Corn-Cob’ Test :
Test Panels
Individually, then
Rotate
Spacecraft

March 3, 2014



1.8kW, Metal-Halide
Lamp System w/ Ballast

Adjustable, 10'
Ratcheting Stand

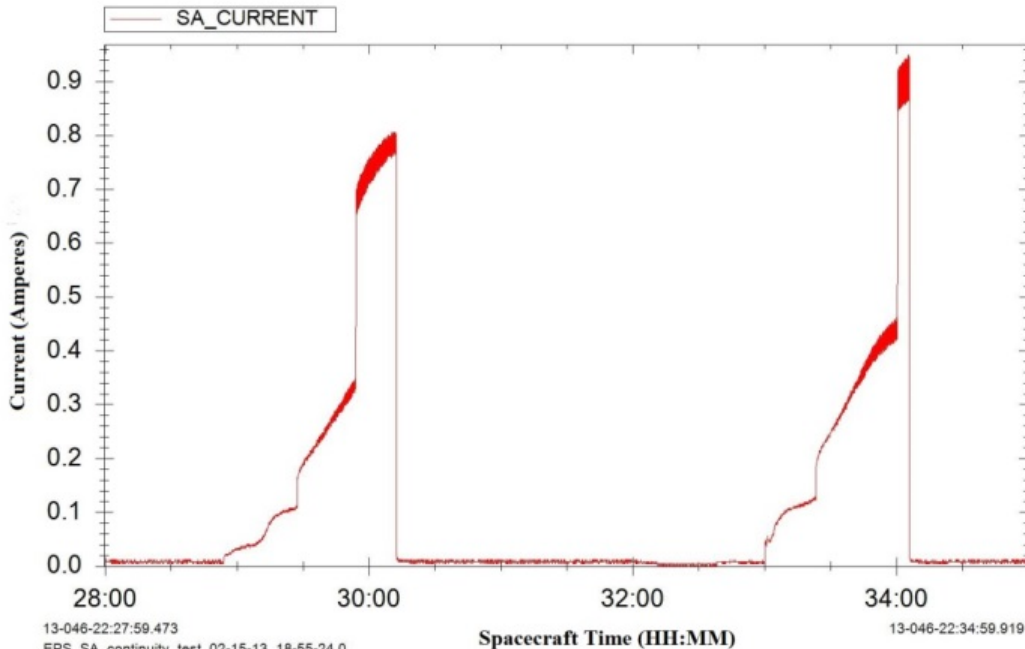


Maintaining System Health

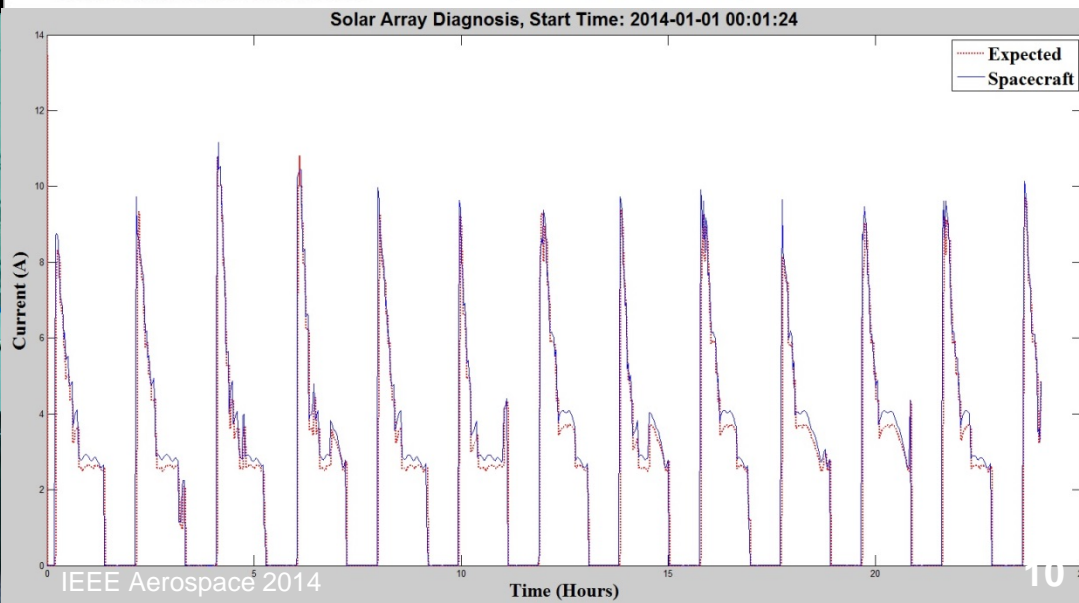
Avionics data capture demonstrates end-to-end functionality & current levels →

Avionics measures only one current for entire array – based on spacecraft attitude, can track each panel's degradation during ops

Launch-site panel inspection yielded 5 cracked cells – repair & retest at launch site!



March 3, 2014





Conclusions

Notes on Class D & Risk:

- Unlikely to have enough resources to guarantee elimination of a risk
- Sometimes it will not be possible to quantify the risk
- Analyze, disposition, accept, and document the risk
- Educated risk taking vs. ignorance
- Sometimes the 'conservative' spaceflight approach is easier – just take the 'high road' because the resources are there

In Conclusion,

- The LADEE risk management process is standard & not novel
- The LADEE EPS team was forced by resources to develop a cost-effective risk solution appropriate for a short-duration mission:
 - Inexpensive (project savings >\$100k)
 - Portable (<100lb)
 - Re-usable – system was brought to & used at the launch site
 - Safe – 'daylight' lamps keep panels cool, \$15 glasses protect eyes



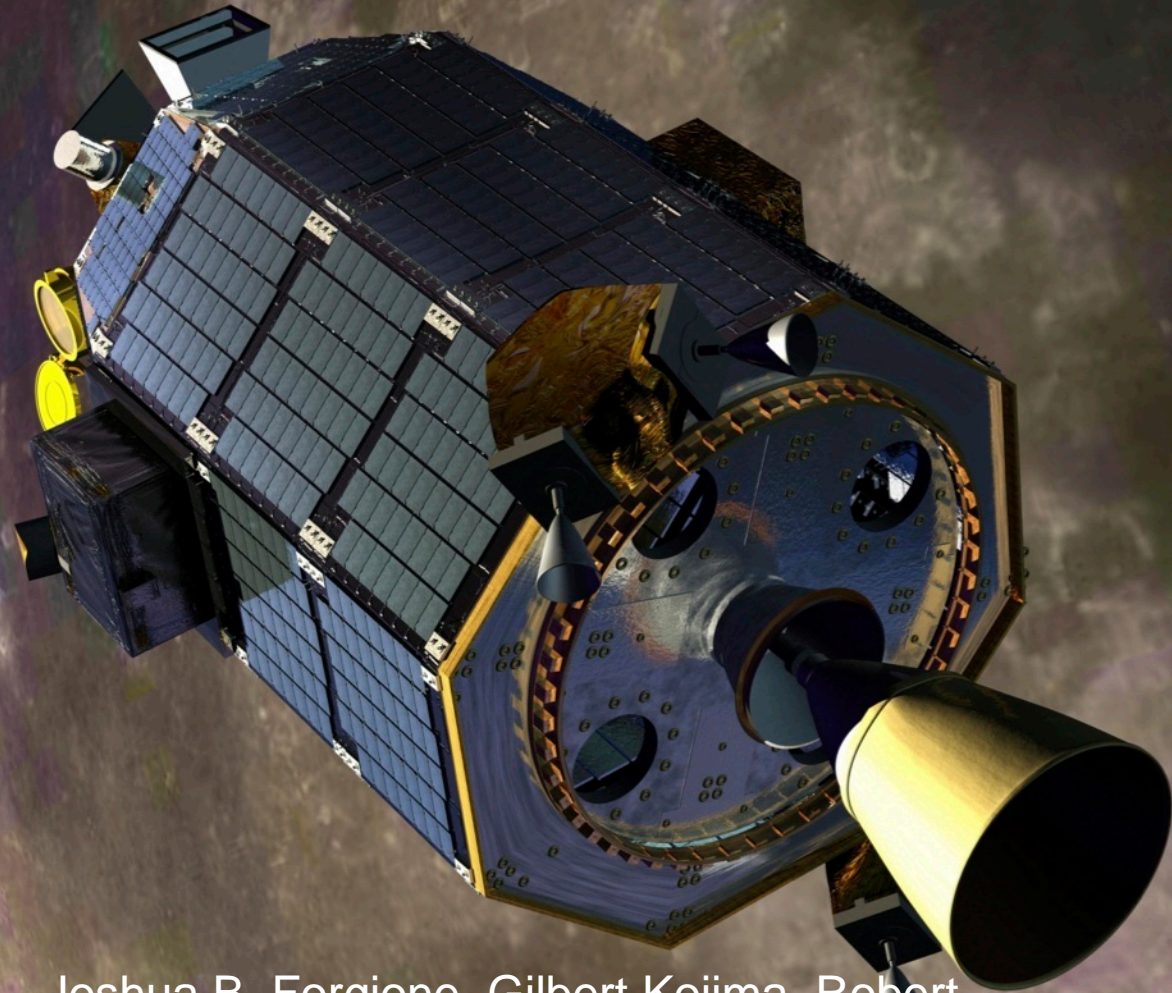
Conclusions

Thank you for your time!

Questions?

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Backup Slides

BACKUP SLIDES!

UV vs. Metal-Halide vs. Sunlight

Spectral Distribution Chart Comparing Different Lighting Technologies

